

IVCHENKO-PROGRESS INNOVATIONS FOR TURBOPROP ENGINES

Sergiy Dmytriyev
Department manager SE Ivchenko-Progress

5th Symposium on Collaboration in Aircraft Design

October 12th-14th, 2015, Naples

“SCIENTIFIC PRODUCTION ASSOCIATION “A. IVCHENKO” CORPORATION



YEAR OF CORPORATION ESTABLISHMENT - 2007

	1907 - Creation of Motor Sich JSC	1945 - Creation of Ivchenko-Progress SE
Sphere of activity	Development, production and overhaul of gas-turbine engines for civil and military aviation, industrial gas-turbine drives and power-generating plants, consumer goods.	
Structure	14 structural units located in different parts of Ukraine as well as in China, India, UAE, Algeria	one structural unit located in Zaporozhye
Number of employees	over 21 000	3 100



CORPORATION 'SPA 'A.IVCHENKO' ON THE MAP

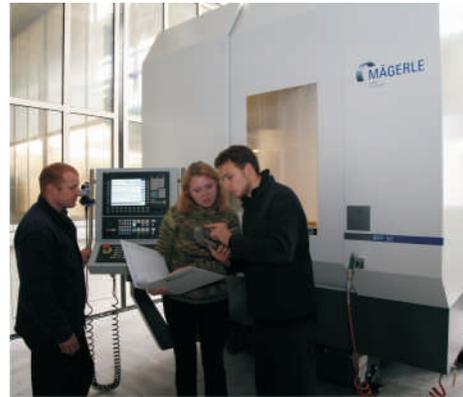


THE BASIC SPHERES OF ACTIVITIES

DESIGN



MANUFACTURE



OVERHAUL



TEST, DEVELOPMENT AND CERTIFICATION



PUTTING IN SERIES PRODUCTION AND IMPROVEMENT OF CONSUMER'S CHARACTERISTICS



DIRECTIONS OF ACTIVITY

CIVIL AVIATION: commercial helicopters and aircraft



STATE AVIATION: trainers and combat trainers, multipurpose aircraft, military transport aircraft and helicopters



Totally 75 certificates of various types



GOSAVIASLUZHBA

**State Department of
Aviation
Transport of Ukraine
Certificates No. SP 004,
No. UA.145.0073, No. 0009 and others**



**Bureau Veritas (France)
Certificate EN 9100:2009 No. FR015515-1,
EN 9110:2009 No. FR015516-1,
ISO 9001:2008 No. UA227484**



**European Aviation
Safety Agency
(Germany)
Certificate No. 216/2008,
No. 1702/2003 Part 21A.23(b)2**

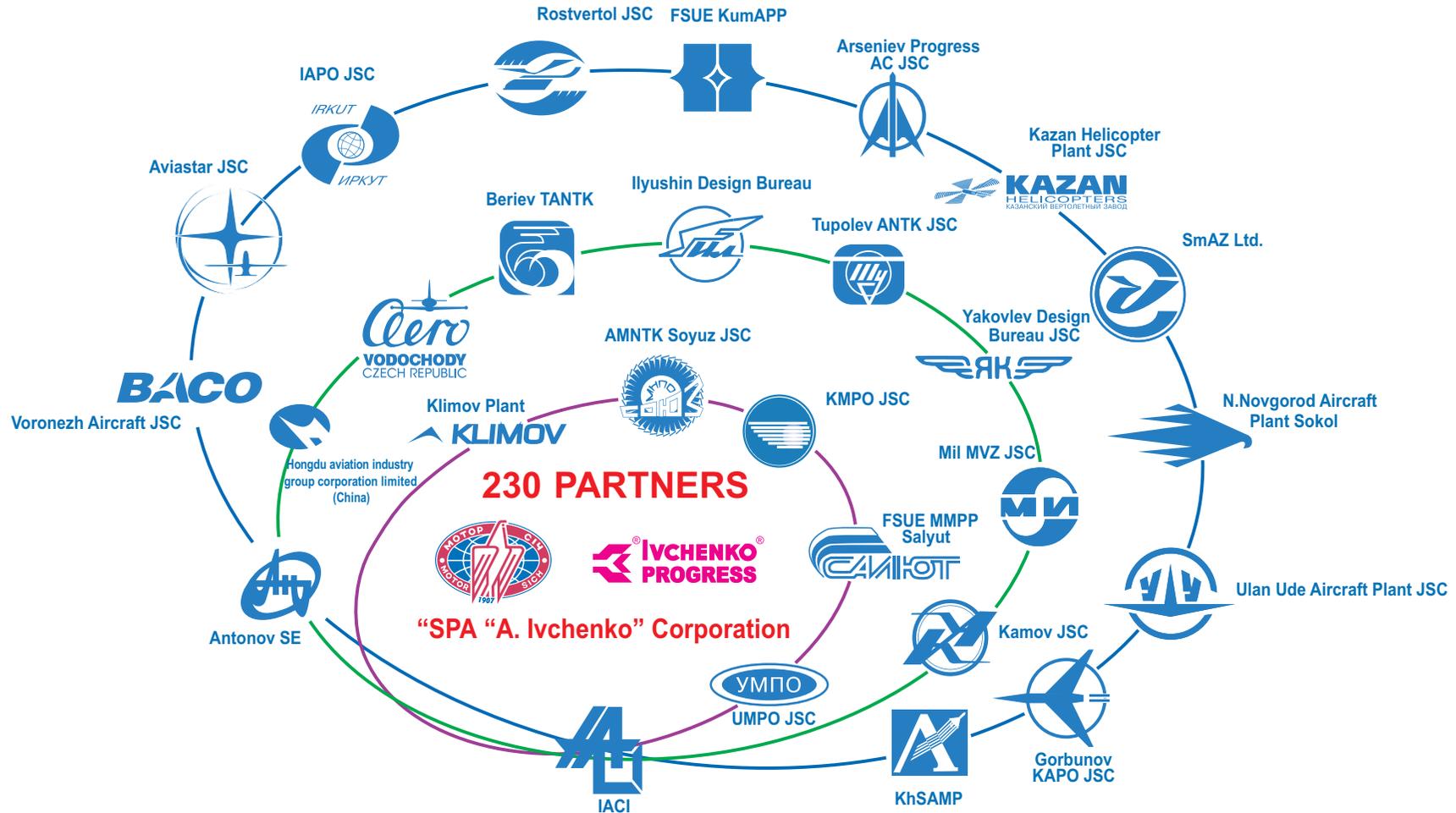


**Aviation Register of
Interstate Aviation Committee
(ARMAK)
Certificates No. SPR-11, No. SPR-15,
No. R-56, No. R-69 and others**



**Federal Air Transport Agency
(Russia)
Certificates No. 2021130360,
No. VR 27.1.4223-2011**

PARTNERS OF “SPA “A. IVCHENKO” CORPORATION DESIGNES AND MANUFACTURERS OF AERO-ENGINES AND AIRCRAFT



Among the partners of “SPA “A. Ivchenko” Corporation are more than **130** designers and suppliers of vendor items and more than **100** suppliers of materials and semi-finished products

COUNTRIES OPERATING AIRCRAFT POWERED BY IVCHENKO CORPORATION ENGINES

IN ALL: 104 COUNTRIES

AZERBAIJAN
ALGERIA
ANGOLA
ARGENTINA
ARMENIA
AFGHANISTAN
BANGLADESH
BELARUS
BULGARIA
BOLIVIA
BRAZIL
BURKINA FASO
BURUNDI
CANADA
CHINA
CHAD
COLUMBIA
CONGO
COTE D'IVOIR
CROATIA
CUBA
CZECH REPUBLIC
CYPRUS
DJIBOUTI
DR CONGO
ECUADOR
EGYPT
EQUATORIAL GUNEA
ERITREA
ESTONIA
ETHIOPIA
GAITI
GEORGIA
GERMANY
GHANA

GREECE
GUINEA-BISSAU
HUNGARY
INDIA
INDONESIA
IRAN
IRAQ
IRELAND
ISRAEL
JAPAN
KAMPUCHEA
KAZAKHSTAN
KENYA
KOREA PDR
KYRGYZSTAN
LAOS
LATVIA
LIBERIA
LIBYA
LITHUANIA
MACEDONIA
MADAGASCAR
MALAYSIA
MALI
MEXICO
MOLDOVA
MONGOLIA
MOZAMBIQUE
NEPAL
NEW ZEALAND
NICARAGUA
NIGER
NIGERIA
PAKISTAN
PALESTINE

PANAMA
PAPUA(NEW GUINEA)
PERU
POLAND
PORTUGAL
ROMANIA
REPUBLIC OF SOUTH AFRICA
RUSSIAN FEDERATION
RWANDA
SENEGAL
SIERRA LEONE
SLOVAKIA
SOMALI
SOUTH KOREA
SPAIN
SRI LANKA
SUDAN
SWITZERLAND
SYRIA
TADJIKISTAN
THAILAND
TURKMENISTAN
TURKEY
WESTERN SAHARA
UGANDA
UKRAINE
UNION OF MYANMAR
UNITED ARAB EMIRATES
USA
UZBEKISTAN
VENEZUELA
VIETNAM
YEMEN
ZIMBABWE

Etc.

Ivchenko-Progress SE Designed Engines

PISTON AND TURBOSHAFT ENGINES FOR LIGHT HELICOPTERS AND ULTRAHEAVY MILITARY TRANSPORT HELICOPTERS



Mi-1



Yak-100



B-10,-11



B-5



G-4

AI-26/GR/GRF
1945*



Ka-26



Ka-18



Ka-15

AI-14V
1951*



Mi-26T



Mi-26



D-136
1971*



Ka-226



Mi-2M



AI-450
1994*



MSB-2



Mi-2M/MSB-2



AI-450M,M1
2006*



Mi-26T2



D-136-2
2010*



RUMAS 245



AI-450M2
2013*



Helicopters of take off weight up to 15 t.



TV3-117VMA-SBM1V series 5,
TV3-117VMA-SBM2V
2013*



* - year of project launch

ENGINES FOR HELICOPTERS

SERIES-PRODUCED AND OVERHAULED ENGINES



AI-450M
modifications



TV3-117VMA-SBM1V
Series 4/ 4E



TV3-117VMA-SBM1V Series 1, 2



D-136



PROJECTS AND NEW ENGINES



AI-450-2



NEW
TURBOSHAFT



TV3-117VMA-SBM1V
Series 5/-SBM2V



AI-8000V



D-136-2/AI-136T



AI-127



TURBOJET BYPASS ENGINES FOR REGIONAL, MILITARY TRANSPORT AND MULTIPURPOSE AIRCRAFT



An-74TK-300



An-74



An-72



Yak-42



M-15



Yak-40



AI-25 Series 2, 2M
1962*



D-36 series 1, 1A,
2A, 3A, 4A
1967*



An-225



An-124



D-18T Series 1, 3
1972*



Be-200



Tu-334



D-436T1,
D-436TP
1991*



An-148, -158



D-436-148
2002*



An-124-100M-150



D-18T
series 3M
2012*



An-178



D-436-148FM
2013*

* - year of project launch

TURBOFAN ENGINES



AI-450BP/BP-2

AI-25

D-36

D-436

AI-28

D-18T/AI-38

4.0 5.5
(410) (560)

14.7
(1 500)

63.8
(6 500)

62.8
(6 400)

84.4
(8 600)

73.6
(7 500)

98.1
(10 000)

229.9
(23 430)

313.9
(32 000)

Takeoff thrust, kN (kgf)

PISTON, BYPASS TURBOJET ENGINES, INCLUDING ENGINES WITH AFTERBURNER, FOR TRAINERS AND COMBAT TRAINERS



L-39M



Yak-130D



Yak-130



L-15



Yak-12



JL-8



L-59(39MS)



AI-222-25F
2005*



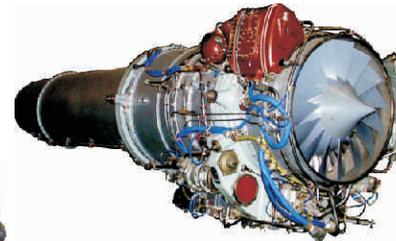
Yak-18



L-39



AI-222-25
1998*



DV-2
1980*



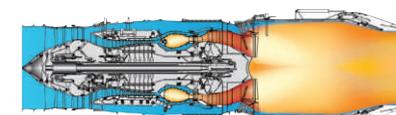
AI-14
1948*



AI-25TL, AI-25TLK,
AI-25TLSh
1970*

* - year of project launch

ENGINES FOR TRAINERS, COMBAT TRAINERS AND LIGHT COMBAT PLANES



AI-25TL/TLK

AI-25TLSH

**AI-222-25
modifications**

**AI-222-30
modifications**

**AI-222
versions with
afterburner**

**AI-9500F
engine with
afterburner**

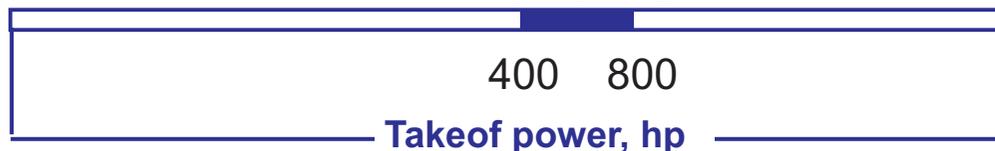
16.9 (1 720)	18.2 (1 850)	24.5 (2 500)	29.4 (3 000)	39.2 (4 000)	49 (5 000)	88.3 (9 000)	98.1 (10 000)
-----------------	-----------------	-----------------	-----------------	-----------------	---------------	-----------------	------------------

Maximum thrust, kN (kgf)

TURBOPROP AND TURBOSHAFT ENGINES



AI-450
turboshaft and
turboprop engines



TURBOFAN ENGINES



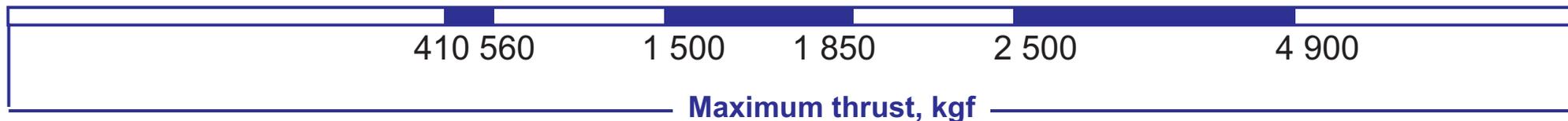
AI-450BP/BP-2



AI-25TL/TLSH



AI-222 modifications



TURBOPROP ENGINES FOR REGIONAL AND MILITARY TRANSPORT AIRCRAFT, SPECIAL PURPOSE AIRCRAFT AND GA



IL-38



IL-20, IL-22



IL-18



An-12



An-10



AI-20A Series 1, 2, 3, 4
AI-20M Series 6
1955*



An-30



An-26



An-24



AI-24 Series 2,
AI-24T, AI-24VT
1957*



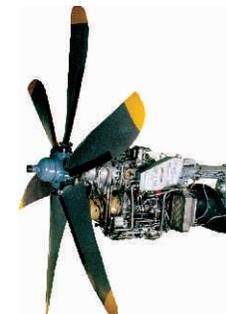
An-70



D-27
1985*



An-140



TV3-117VMA-SBM1
1996*



EV-55



AI-450S2
2011*



DA50-JP7



AI-450S
2012*



* - year of project launch

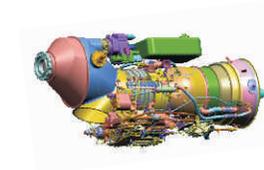
TURBOPROP AND TURBOPROPFAN GAS-TURBINE ENGINES



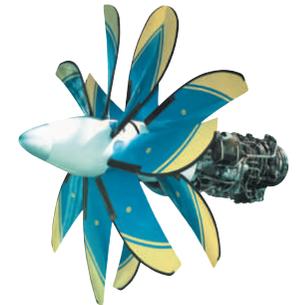
AI-450S/S-2



TV3-117VMA-SBM1/2 AI-40



AI-800



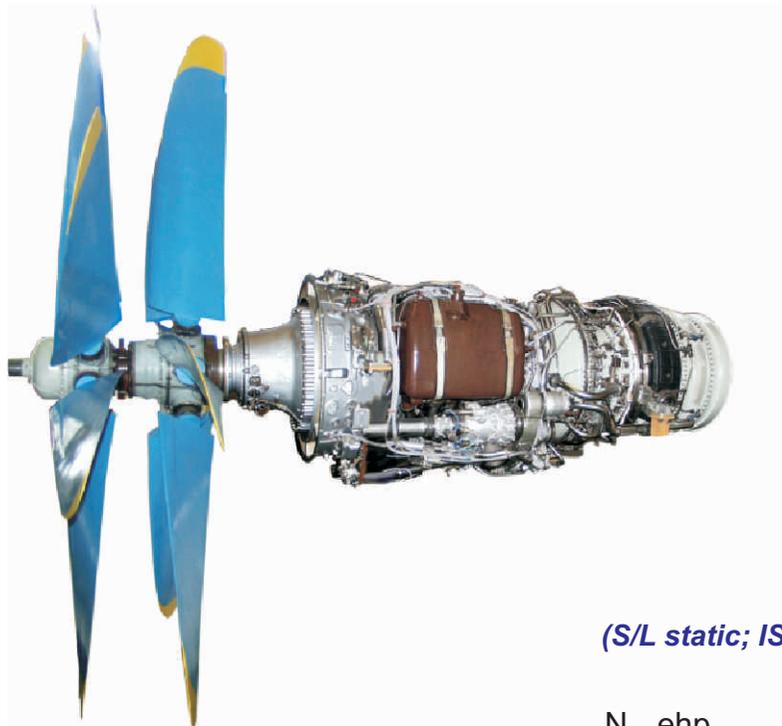
D-27

465	800	2 500	3 600	3 600	4 000	7 600	8 300	13 240
(340)	(590)	(1 840)	(2 650)	(2 650)	(2 940)	(5 588)	(6 102)	(9 735)

Takeoff power, hp (kW)

Innovations for Turboprop Engines

Advanced contra-rotating (open rotor design) engine



(S/L static; ISA)

N_e , ehp	13 240
C_N , kg/h/ehp	0.180

D-27

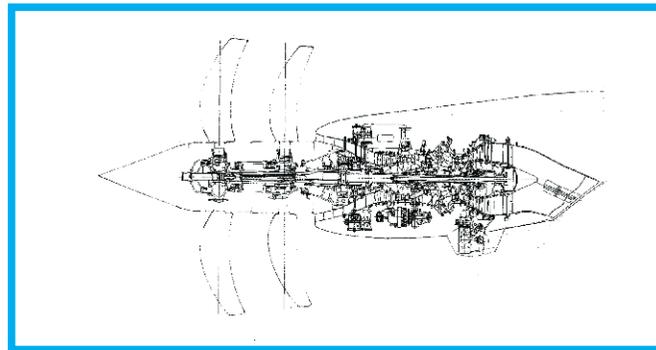
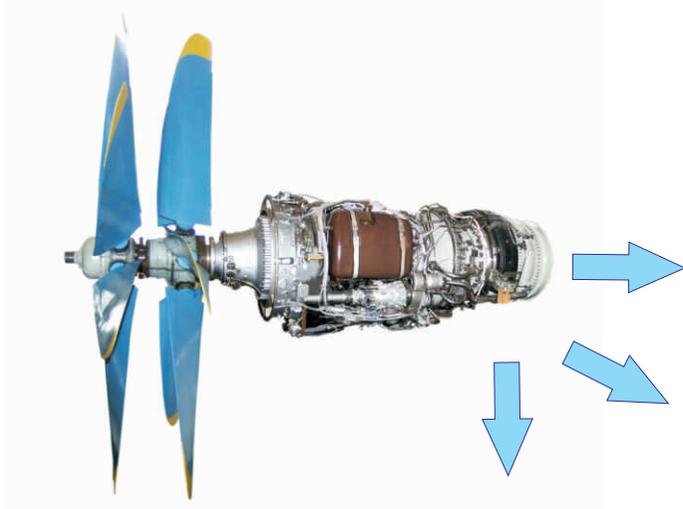
Takeoff

An-70

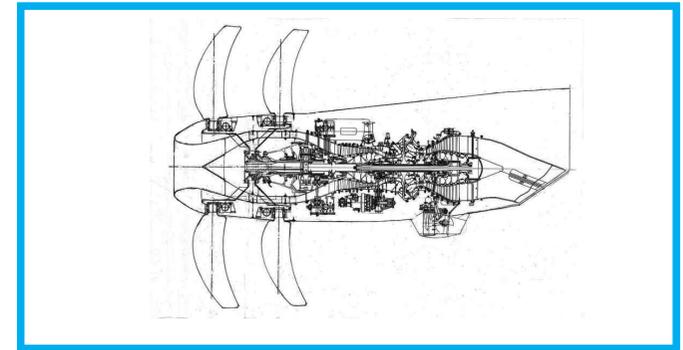


At present activities for mastering the D-27 propfan engine serial production are carried out

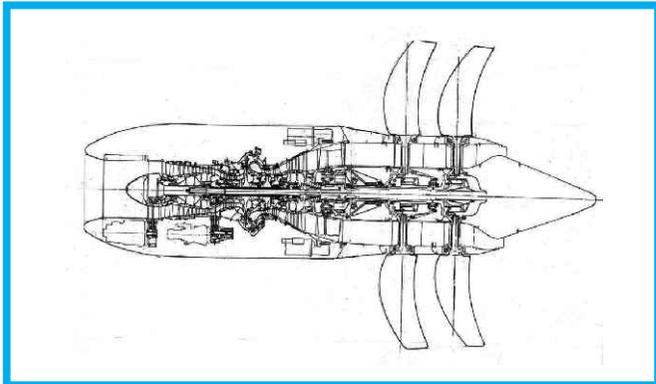
CREATION OF OPEN-ROTOR TYPE ENGINE DEMONSTRATOR



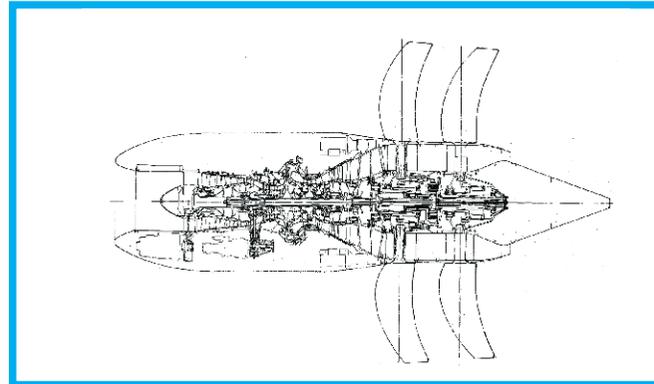
Tractor propfan of 3.5 m in diameter with retrofitted air intake



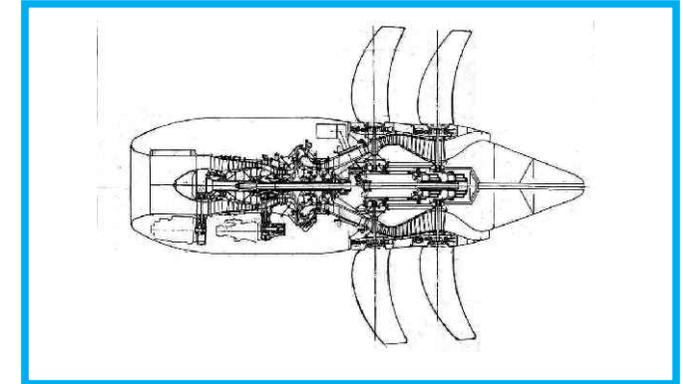
Tractor propfan with forward-type air intake



Pusher propfan with propfan-rotor front reduction gear



Pusher propfan with reduction gear between propfan rotors



Pusher propfan with direct propfan drive

AI-450S, AI-450S-2

Engine can be used on similar aircraft



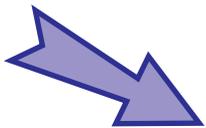
AI-450S



DA50-JP7 - 19/01/2015 Maiden flight



UAV



Light multi-purpose aircraft



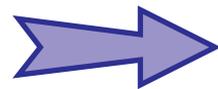
Light trainers



(S/L static; ISA)	AI-450S	AI-450S-2
N_{prop} , hp	450...495	630...800
C_N , kg/h/hp	0.277	0.259

Takeoff

AI-450S-2



General-aviation light aircraft



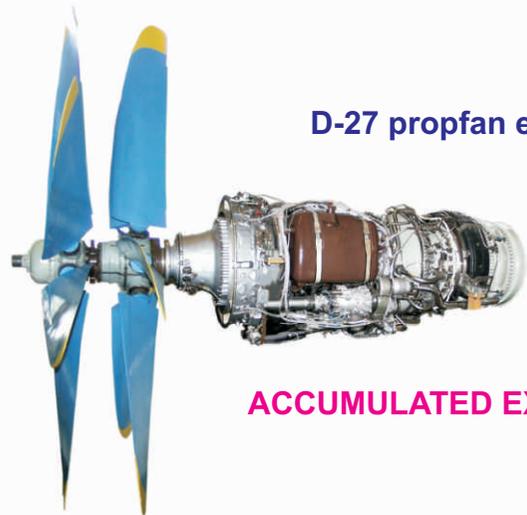
Light multi-purpose aircraft



Developed under **ESPOSA** project of European 7th Framework Programme

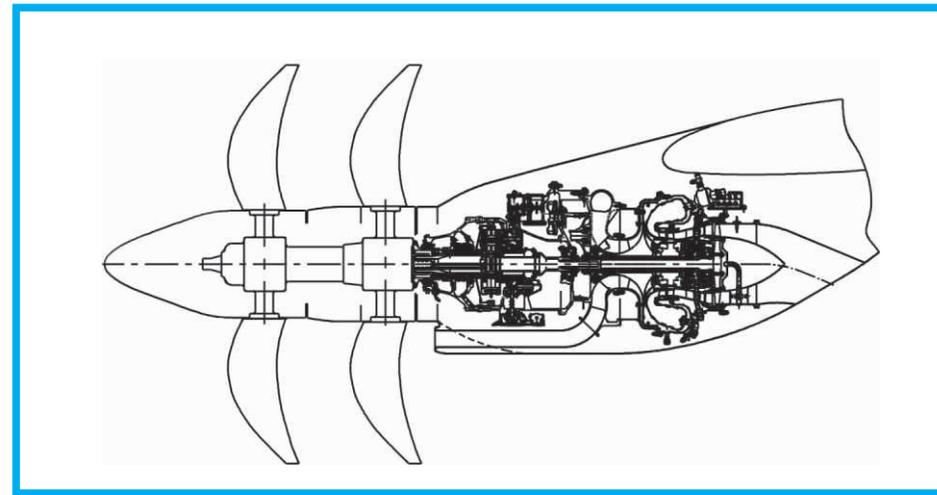
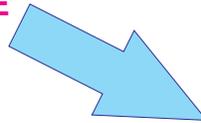
DEVELOPMENT OF ADVANCED GTEs FAMILY CONCEPT (OPEN ROTOR) TO POWER LIGHT AIRPLANES

450...750 hp



D-27 propfan engine

ACCUMULATED EXPERIENCE

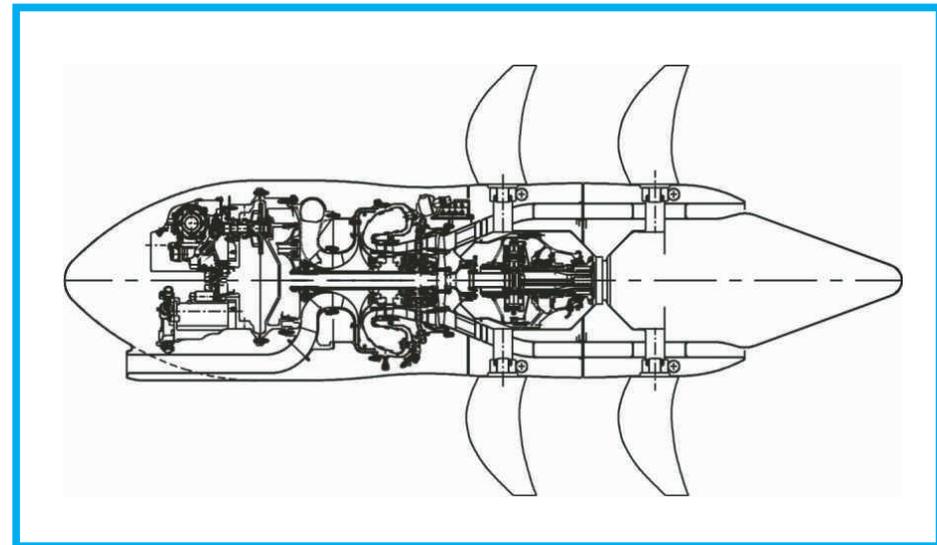
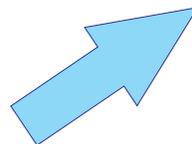


Tractor propeller



AI-450S/S-2 turboprop engine

ADVANCED CORE



Pusher propeller

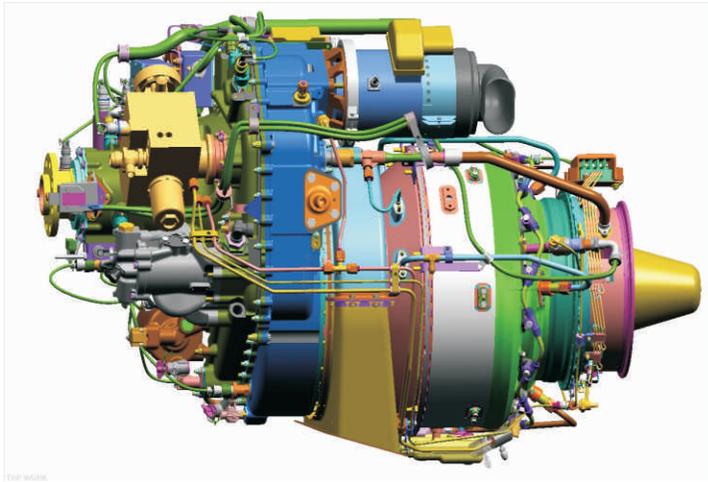
Core developed under **ESPOSA** project
of European 7th Framework Programme

ESPOSA - Efficient Systems and Propulsion for Small Aircraft



ESPOSA

Starting date - 01/10/2011. Duration in months - 57



Role in the project

SE Ivchenko-Progress will develop the mathematical model and perform trade-off studies for estimation of the performance data for the baseliner engine 2 (BE2) configuration, fulfil the specification and optimization of thermodynamic parameters of BE2 engine in the aircraft-enginepropeller system, develop a design of technical requirements for units and engine systems of BE2 and a design of laws of BE2 engine management, **find an optimal design solution for high pressure compressor, realize the optimization, design and experimental investigations of high efficiency advanced small turbine, fulfil advanced dynamic modelling of high speed turbomachine, supply the specifications of the turbine components coating and carry out the demonstrator validation in terms of coating quality and performance under real service conditions, provider of complex tests and validation of BE2 in test rig, verify mathematical model of BE2 engine with results of tests, investigation of capacity development efficiency of GTE family designs for small aircraft on the basis of single advanced turbocompressor.**



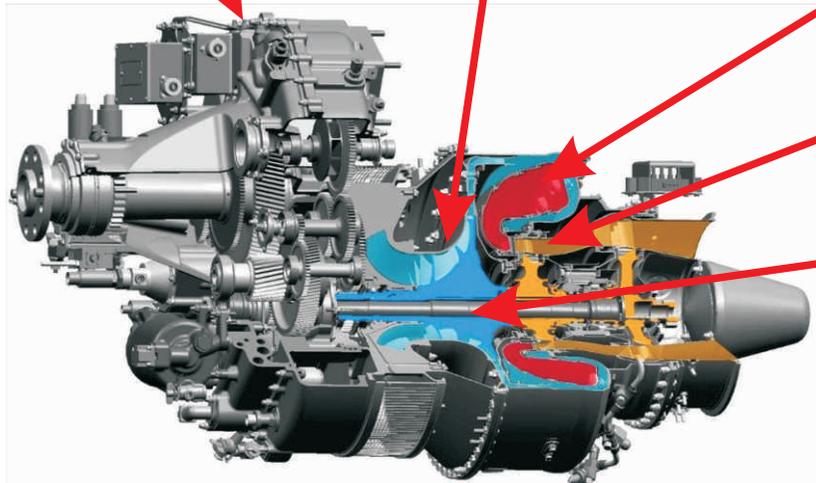
Advanced automatic control system for small engines

**Optimal small compressor
(Advanced centrifugal compressor impeller with a new blade geometry- 'winglet')**

**Efficient combustion concept
(High efficiency, low emission combustor with new designed fuel injector with swirlers)**

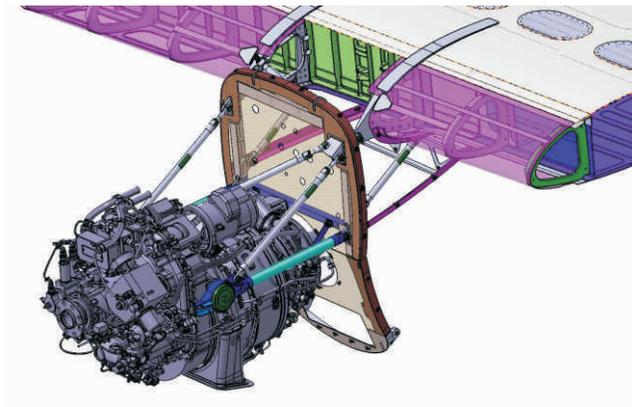
**Advanced Cooled Small Turbine
(15 mm size of airfoil)**

Advanced dynamic modelling of high speed turbomachinery

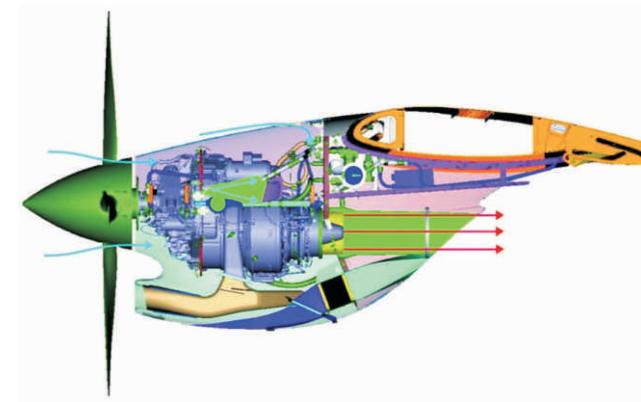




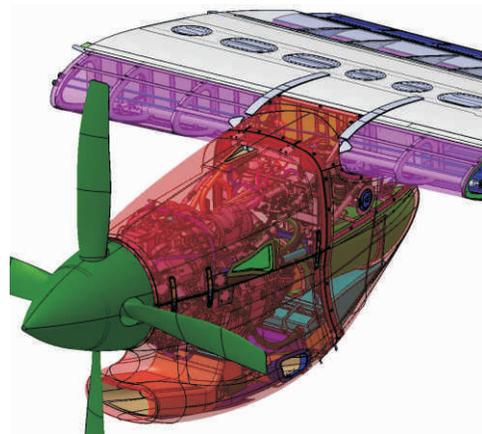
Complex design methodology for Engine Mechanical Integration



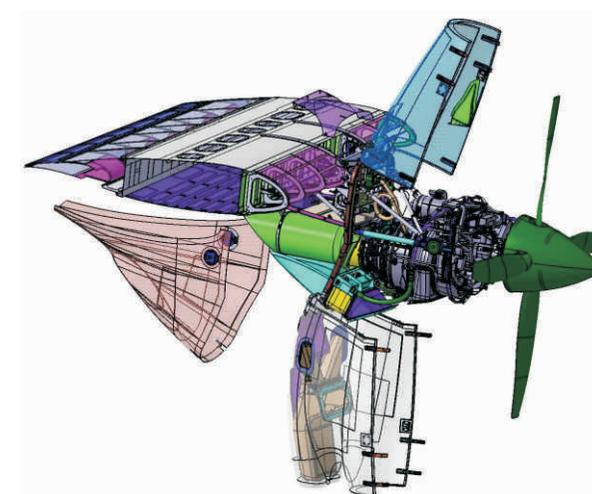
Engine Nacelle Ventilation. Reliable simulation tools for engine thermal integration



Reliable design methodology for aerodynamic and acoustic engine/airframe integration



NACELLE. Good accessibility and maintainability



Technology Innovations for Turboprop Engines

DEVELOPMENT OF PROGRESSIVE COATING SOLUTIONS FOR ENGINE PARTS



Wear resistant coatings for bearings mounting seats on engine shafts



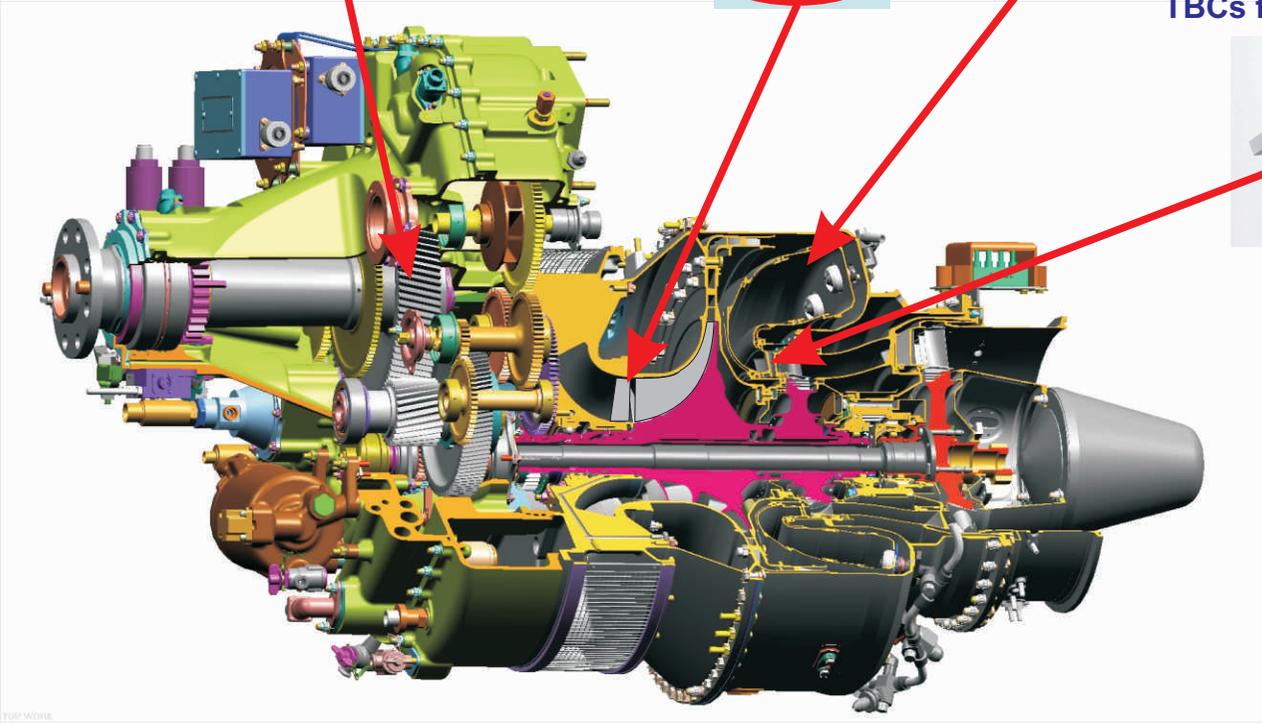
Heat-protective coating for internal surface of small combustor



Anti-fretting and anti-wear coatings for compressor rotor blade roots



TBCs for turbine nozzle guide vanes



OXIGEN - Oxide Dispersion Strengthened Materials for the Additive Manufacture of High Temperature Components in Power Generation



Starting date - 01/02/2013. Duration in months - 48

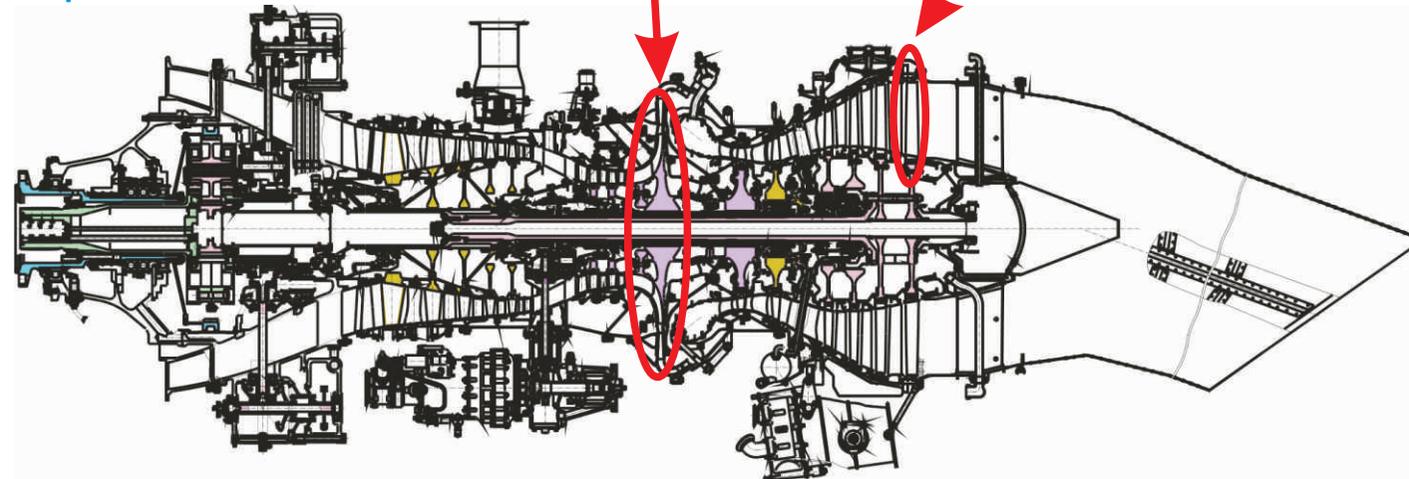
The manufacture of a compressor wheel and turbine blade ODS titanium-aluminium alloy will allow operation at $>800^{\circ}\text{C}$ (theoretically, based on increased high temperature creep strength properties) which would correspond to an increase in operating (inlet/outlet) temperatures by $200-300^{\circ}\text{C}$

Application of new materials will give following effects: reduction of specific fuel consumption of more than 6 %, increase of power can be reached of more than 15 %, weight reduction, engine life increasing, giving a great positive effect on the engine in whole

ODS TiAl alloy

Centrifugal compressor wheel

Turbine blade

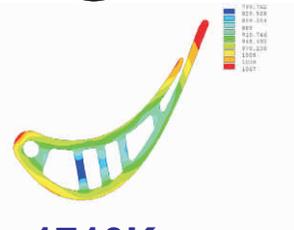
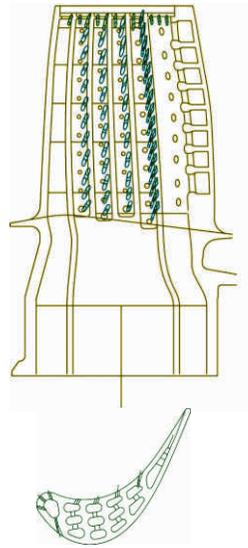
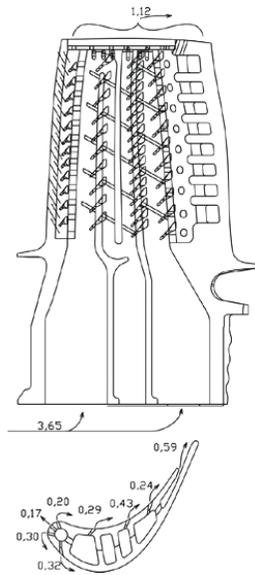
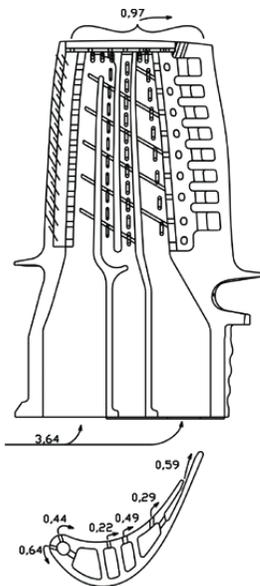


ADVANCED AERO-ENGINE HIGH PRESSURE TURBINE BLADE COOLING SYSTEM CONCEPT

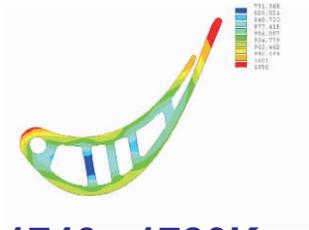
**Modification 1,
modified blade with
shaped film cooling holes**

**Modification 2,
wall-cooled system
("penetrating cooling")**

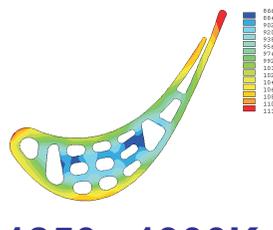
Base design



1710K



1740...1780K



1850...1900K



Blade Mean Section temperatures	Base	Modification 1	Modification 2
Leading Edge, °C	1067	1050	1031
Trailing Edge, °C	1063	1045	1015
Bulk Metal, °C	943	920	875

Modification 2, wall-cooled system blades has an average temperature of the middle section 68°C lower than the blade with base design.

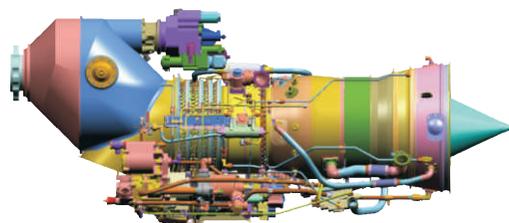
Additionally, the use of TBC with 0.14 mm thickness and a thermal conductivity 2.8 W/(m•K) leads to the decrease in the average temperature of the blade cross section for 15 ... 20°C for all presented cooling blades.

The results of presented investigation show, that one of the most challenging cooling system for the engines with gas temperature at HPT rotor blade inlet TET = 1800...1900 K is a wallcooled system ("penetrating cooling")

New Designed Engines

TV3-117VMA-SBM2, AI-40, AI-8000

TV3-117VMA-SBM2



(S/L static; ISA)

	Takeoff	Emergency
N_e , hp	2 800	3 600
flat-rated to $t^{\circ}\text{C}$	+35	+25
C_e , kg/h/ehp	0.210	

TV3-117VMA-SBM2

MA60/ MA700 type
Regional airplanes



AI-40



(S/L static; ISA)

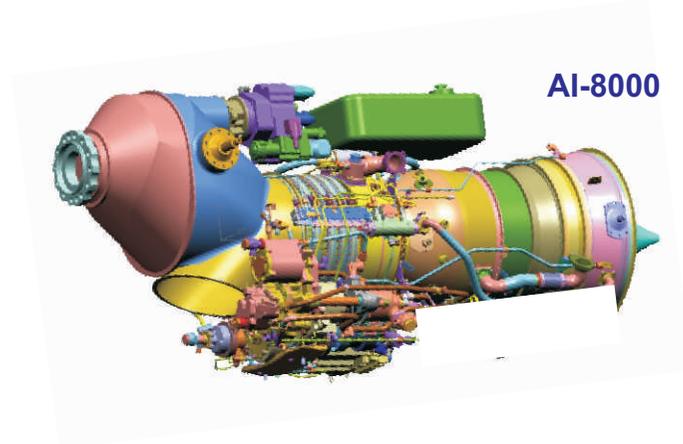
	Takeoff	Emergency
N_e , ehp	3 600	4 000
flat-rated to $t^{\circ}\text{C}$	+40	+40
C_e , kg/h/ehp	0.196	

AI-40

Regional passenger
and light transport aircraft



AI-8000



(S/L static; ISA)

	Takeoff	Emergency
N_e , ehp	7 600	8 300
C_e , kg/h/ehp	0.175	

AI-8000

Transport and passenger aircraft



OUR MAIN GOALS FOR CUSTOMER SUCCESS

Safety

Reliability

Innovation



Economy

Efficiency

Modernization

We provide power to your fly



ENGINES DESIGNED BY IVCHENKO CORPORATION HAVE BEEN OPERATED IN MORE THAN 100 COUNTRIES OF THE WORLD



Zaporozhye Machine-Building Design Bureau Progress
State Enterprise named after Academician A.G. Ivchenko

2, Ivanova Str.,
69068, Zaporozhye, Ukraine
Tel.: +38 0(612) 65-03-27;
Fax +38 0(612) 12-89-22, 65-46-97
progress@ivchenko-progress.com
www.ivchenko-progress.com



THANK YOU

FOR ATTENTION!